

# A Mechanically Pumped Two-Phase Ammonia Fluid Loop for Thermal Control

Ben Furst, Thermal Fluids Group Team: Stefano Cappucci, Takuro Daimaru, Eric Sunada



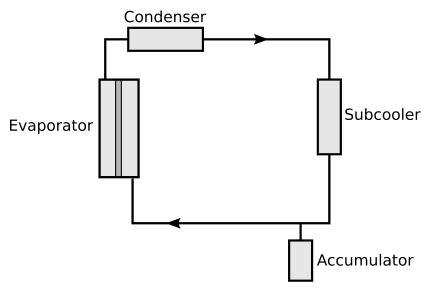
#### **Overview**

- The JPL Two-Phase Technology Group has developed a novel mechanically pumped two-phase fluid loop for thermal control
- Architecture is based on a modified Capillary Pumped Loop (CPL)
- A fully operational testbed using the target flight fluid (ammonia) has been built and tested
  - Test results demonstrate that the system is feasible

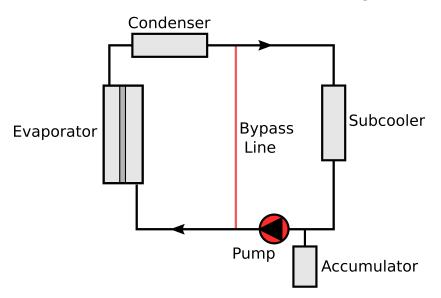
# **System Architecture**

- Architecture is based on a Capillary Pumped Loop (CPL)
  - Additions to CPL include:
    - 1. A mechanical pump,
    - 2. A bypass line
    - 3. An additively manufactured planar evaporator

#### **Typical CPL Architecture**

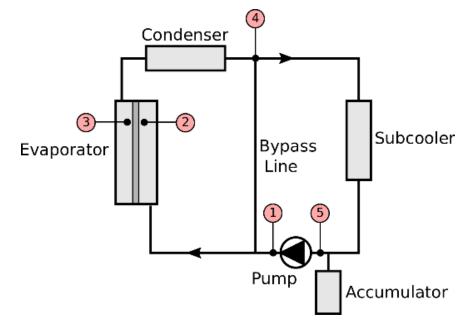


#### **CPL with Mechanical Pump**



# **System Operation I**

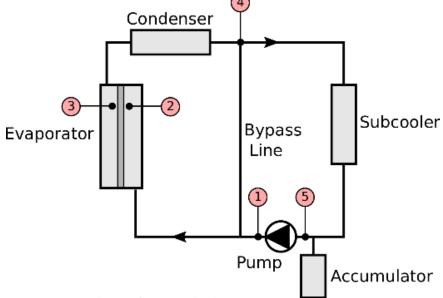
What does the pump do?



- Assume steady operation with meniscus established in evaporator
  - Liquid and vapor are separated at meniscus (P<sub>3</sub> > P<sub>2</sub>)
  - Flow is single-phase everywhere except in condenser
- The pump does not push liquid through evaporator wick
  - Meniscus behaves like a hydrodynamic wall since at meniscus: P<sub>vapor</sub> > P<sub>liquid</sub>
- The pump only pushes liquid through the bypass line
  - This allows the pressure at the condenser outlet (P<sub>4</sub>) to be dictated by the pump flowrate and pressure drop in the bypass line

# **System Operation II**

How does the pump help?



Consider a pressure balance between point 1 and 4

$$\Delta P_{1,4} = \Delta P_{1,2} + \Delta P_{3,4} - \Delta P_{3,2}$$

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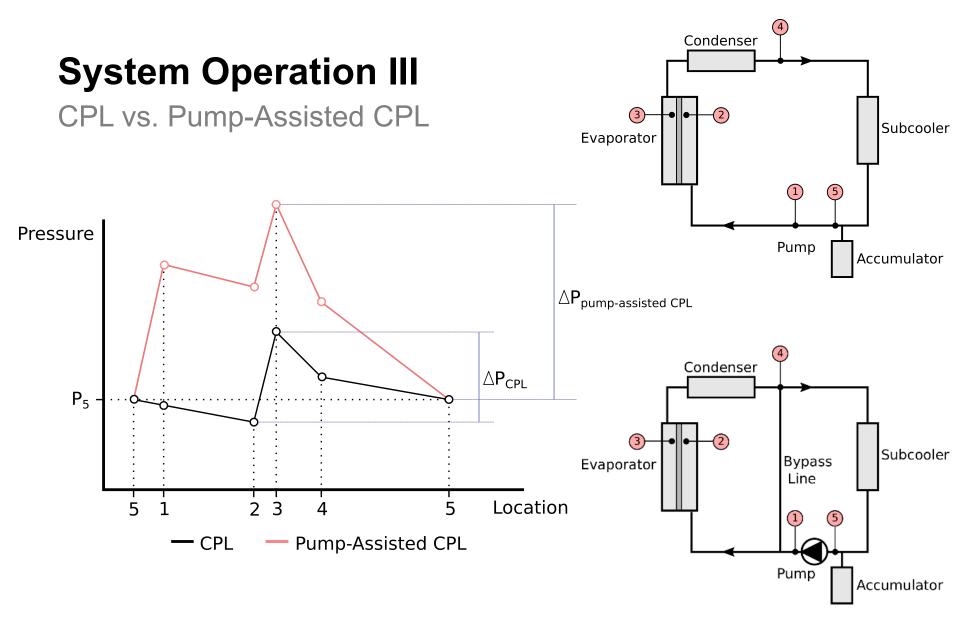
$$\Delta P_{1,4} = \Delta P_{1,4}$$

$$\Delta P_$$

• Solve for capillary pressure:  $\Delta P_{3,2}$ 

$$\Delta P_{3,2} = \Delta P_{1,2} + \Delta P_{3,4} - \Delta P_{1,4}$$

- Capillary pumping is assisted by pressure drop in bypass line
- > The mechanical pump covers the pressure drop in the bypass line



Addition of the mechanical pump can significantly increase pumping capacity

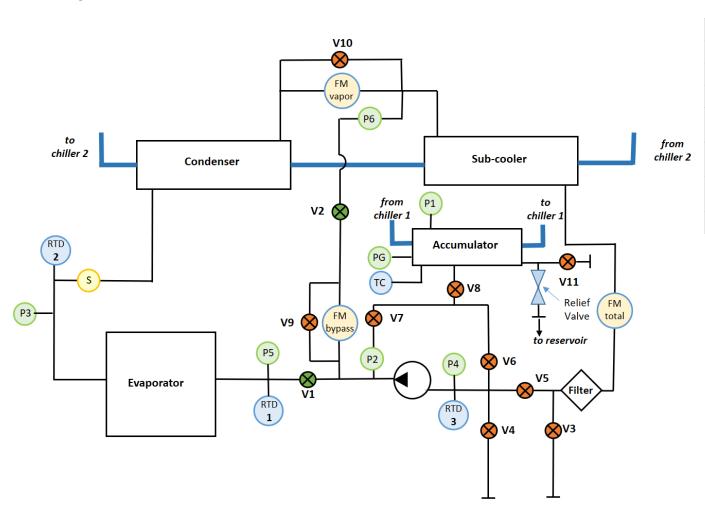
# **Advantages of Pump-Assisted CPL Achitecture**

- Adds additional capability to the classic CPL architecture
  - Can accommodate larger pressure drops due to pump
    - Higher heat loads are possible
    - Longer transport lengths possible
  - Simplifies integration and testing
    - Can incorporate mechanical fittings/valves
    - Less sensitive to adverse orientations during ground testing
  - More robust operation
    - Mechanical pump gives additional control authority
    - Could operate as a passive CPL with degraded performance if pump fails

# **Testing Overview**

- A pump-assisted CPL has been built and is currently under test
  - Developed over past 3 years
- Currently working with an operationally flight-like system
  - Working fluid: Ammonia (target flight-fluid)
  - Incorporates all major system components in actual configuration
  - Instrumented to monitor temperature, pressure, flowrate
- System has demonstrated stable, repeatable performance
  - System is operating as anticipated
  - Over 350 hours of testing completed
- Recent test campaign showed promising results
  - Stable transport of heat loads from 30 W to 850 W
  - Heat fluxes sustained up to 13 W/cm<sup>2</sup>
  - Maintained isothermal planar evaporator (± 2°C) between 30 W and 300 W for a fixed pump speed

# System Schematic

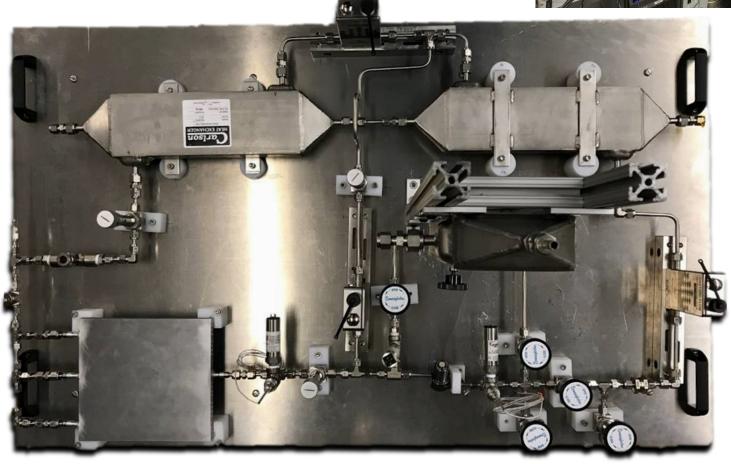


| Operating Specifications                                       |                        |  |
|--|------------------------|--|
| Fluid  | Ammonia                |  |
| Nominal operating temperature                                  | 20°C                   |  |
| Max operating temperature                                      | 30°C                   |  |
| Nominal operating pressure<br>(Ammonia vapor pressure at 20°C) | 124 psia<br>(110 psig) |  |
| Max planned working pressure (Ammonia vapor pressure at 30°C)  | 170 psia<br>(155 psig) |  |
| Relief valve set pressure                                      | 215 psia<br>(200 psig) |  |
| System Proof Pressure  | 250 psig               |  |

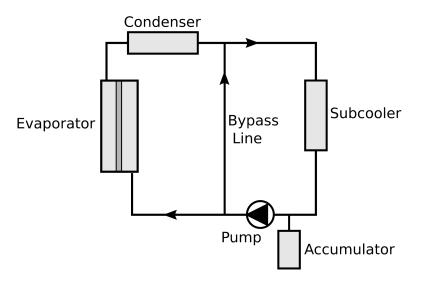
| Legend |  |                          |
|--------|--|--------------------------|
| P      |  | Press. transducer (abs.) |
| PG     |  | Pressure gauge (analog)  |
| T      |  | Temperature gauge        |
| TC     |  | Thermocouple             |
| FM     |  | Flow meter               |
| S      |  | Sight glass              |
| 8      |  | Ball valve               |
| 8      |  | Needle valve             |

Hardware



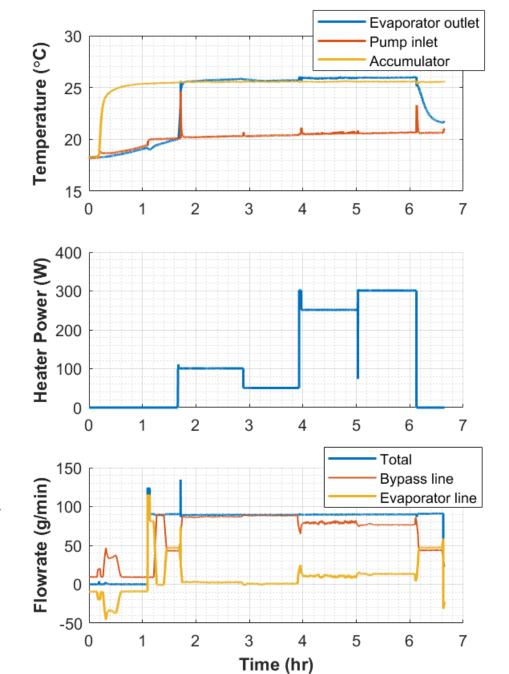


**Experimental Data I** 

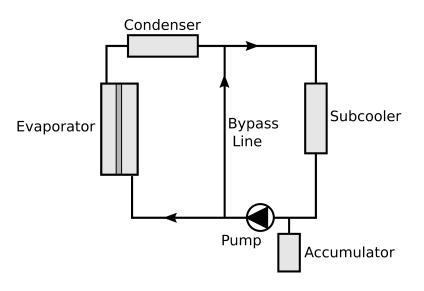


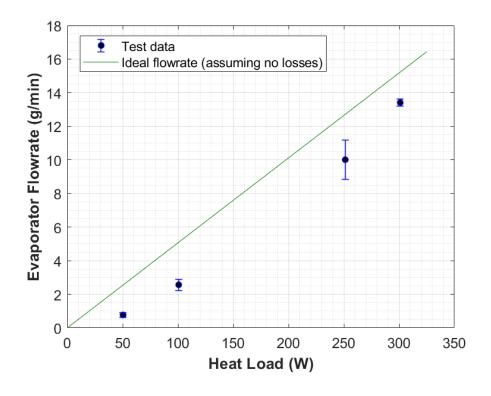
#### **Notes**

- Total flowrate fixed @ 90 g/min
- Heat load varied from 50 W to 300 W
- Temperature @ evaporator outlet steady at ~27°C
- As heat load increases, evaporator flowrate increases



**Experimental Data II** 





#### **Notes**

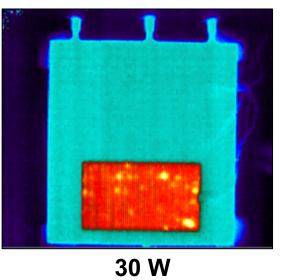
- As the heat load increases, the flowrate through the evaporator increases
- This implies that
  - a) The evaporator wick is working as a capillary pump
  - b) The fluid phases are separated with pure vapor only existing between evaporator and condenser
- The system is working as anticipated

Evaporator IR Images

Pump Flowrate: 90 g/min

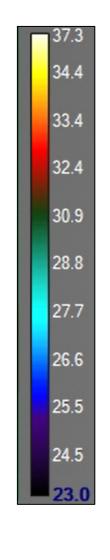
Accumulator Temperature: 26°C

**Evaporator Temperature:** 28°C



150 W

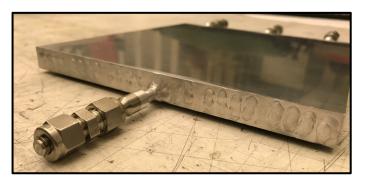
100 W

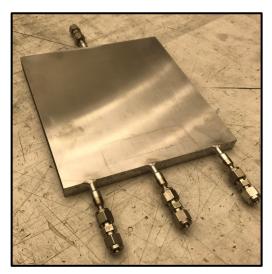


(100 W top; 50 W bottom) (1

325 W (150 W top; 175 W bottom)

**Evaporator Design** 





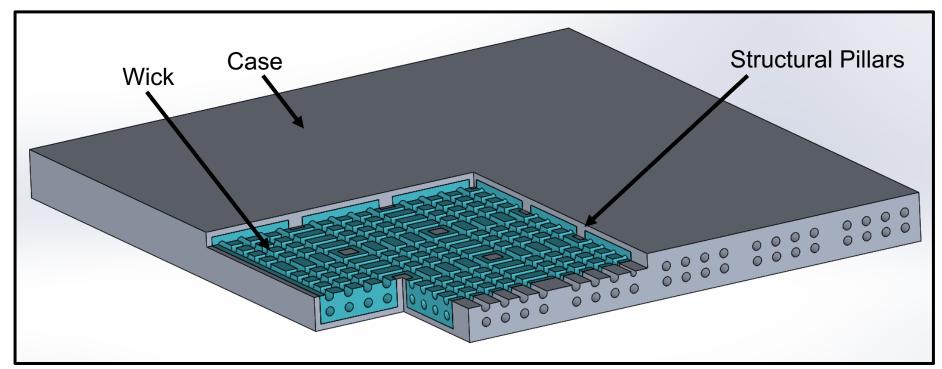
**Fabrication:** DMLS **Material:** Aluminum

**Size:** 8.4" x 7.8" x 0.63"

MAWP: 200 psig

Max Pore Size: 22 μm Permeability: 1e-13 m<sup>2</sup>

Porosity: 24%



## Conclusion

- A new architecture for a pump-assisted CPL has been developed
- A prototype ammonia testbed has been built and tested
  - System incorporates a novel AM planar evaporator
- Preliminary test results indicate that the system is feasible
  - System operated as expected
  - Transported heat loads from 30 W to 850 W
  - Max heat flux: 13 W/cm<sup>2</sup>
  - Subcooling demonstrated from 3°C 10°C

#### **Future Work**

- Refine evaporator design
  - Reduce thickness and increase effectiveness
- Increase TRL of system
  - Integrate flight-like components into testbed
  - Continue to experimentally characterize system
  - Develop analytical/numerical design and prediction capability
  - Purse flight-demo opportunities

### References

- 1. Furst, Benjamin, et al. "A Comparison of System Architectures for a Mechanically Pumped Two-Phase Thermal Control System." 47th International Conference on Environmental Systems, 2017.
- 2. Cappucci, Stefano, et al. "Working Fluid Trade Study for a Two-Phase Mechanically Pumped Loop Thermal Control System." 48th International Conference on Environmental Systems, 2018.
- 3. Furst, Benjamin, et al. "An Additively Manufactured Evaporator with Integrated Porous Structures for Two-Phase Thermal Control." 48th International Conference on Environmental Systems, 2018.
- Sunada, Eric, et al. "A two-phase mechanically pumped fluid loop for thermal control of deep space science missions." 46th International Conference on Environmental Systems, 2016.



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